

## DETERMINATION LHP OF AXISYMMETRIC TRANSIENT MOLYBDENUM STEEL-4037H QUENCHED IN SEA WATER BY DEVELOPING 1-D MATHEMATICAL MODEL

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### Abstract

The modelling of an axisymmetric industrial quenched molybdenum steel bar AISI-SAE 4037H quenched in sea water based on finite element method has been produced to investigate the impact of process history on metallurgical and material properties. Mathematical modelling of 1-dimensional line (radius) element axisymmetric model has been adopted to predict temperature history and consequently the hardness of the quenched steel bar at any point (node). The lowest hardness point (LHP) is determined. In this paper hardness in specimen points was calculated by the conversion of calculated characteristic cooling time for phase transformation  $t_{8/5}$  to hardness. The model can be employed as a guideline to design cooling approach to achieve desired microstructure and mechanical properties such as hardness. The developed mathematical model was converted to a computer program. This program can be used independently or incorporated into a temperature history calculator to continuously calculate and display temperature history of the industrially quenched steel bar and thereby calculate LHP. The developed program from the mathematical model has been verified and validated by comparing its hardness results with commercial finite element software results. The comparison indicates reliability of the proposed model.

*Key words: Heat Treatment; Quenching; Axisymmetric Molybdenum Steel Bar; Finite Element; Mathematical Modeling; Unsteady State Heat Transfer.*

### Introduction

Quenching is a heat treatment usually employed in industrial processes in order to control mechanical properties of steels such as hardness [1, 2]. The process consists of raising the steel temperature above a certain critical value, holding it at that

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temperature for a specified time and then rapidly cooling it in a suitable medium to room temperature [1,3]. The resulting microstructures formed from quenching (ferrite, cementite, pearlite, upper bainite, lower bainite and martensite) depend on cooling rate and on chemical composition of the steel [1, 4].

Quenching of steels is a multi-physics process involving a complicated pattern of couplings among heat transfer. Because of the complexity, coupled (thermal-mechanical-metallurgical) theory and non-linear nature of the problem, no analytical solution exists. However, numerical solution is possible by finite difference method, finite volume method, and the most popular one - finite element method (FEM) [1, 5].

During the quenching process of the steel bar, the heat transfer is in an unsteady state as there is a variation of temperature with time [1, 6]. In this paper the heat transfer analysis will be carried out in 3- dimensions. The three dimensional analysis will be reduced into a 1-dimensional axisymmetric analysis to save cost and computer time [1, 5, 7-10]. This is achievable because in axisymmetric conditions the temperature deviations is only in r-direction while there is no temperature variation in the theta  $\theta$  and z-direction as it is clear in Figs. 1-3. The Galerkin weighted residual technique is used to derive the mathematical model. In this work, 1-dimensional line (radius) element will be developed.

It is clear that the first point (node) will be completely cooled after quenching (surface node) because it is located on the surface touched by the cooling medium, then the other points (nodes) on the radial axis to the centre accordingly will be cooled and the last point will be completely cooled after quenching (centre node)[1].

It means that the maximum hardness will be measured on the surface node subjected to fast cooling, then the hardness will decrease from the surface node on the radial axis to the centre node of the quenched steel bar, respectively. This means that the lowest hardness point of the quenched steel bar will be detected at the centre node [1].

The lowest hardness point (LHP) should be expected inside the heat treated quenched steel bar at the half of the length at the centre of the bar (centre node). To prove this statement experimentally is an almost impossible task using manual calculation techniques. Also the earlier methods only used hardness value calculated at the surface (surface node), which is higher than the lowest hardness point (centre node). This might have negative consequences resulting to the deformation and failure of the component [1].

It will be more important to know the LHP (centre node) when the radius of the quenched steel bar will increase because the lowest hardness point will be lower than the hardness on the surface (surface node). This means that increasing the radius of the bar is inversely proportional to LHP (centre node), while the hardness at the surface (surface node) will be the same [1].

No published informations are available till date on this aspect. This paper represents a contribution towards understanding of steel behaviour at elevated temperature during quenching at the LHP (centre node) of the steel bar. We believe that the results of this paper might be very useful to obtain the hardness of the lowest point of the steel bar in order to reach the maximum benefit against the deformation and failure of the component [1].

### Mathematical model

The temperature history of the quenched cylindrical molybdenum steel bar at any point will be calculated. Three dimensional heat transfers can be analyzed using one dimensional axisymmetric element as shown in Figs. 1-3. [34-39].

#### *Methodology of building the Finite Element (FE) Model in details*

The temperature distribution inside the cylindrical steel bar when reached thermal equilibrium will be calculated. There are special classes of three-dimensional heat transfer problem:

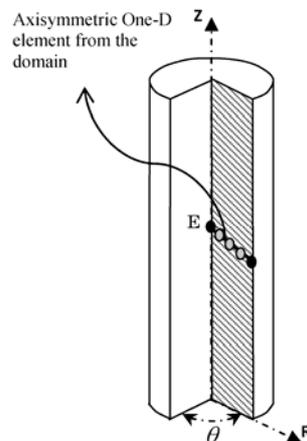
- i. geometrically axisymmetric.
- ii. each thermal load is symmetrical about an axis.

This 3-dimensional heat transfer problem may be analyzed using one-dimensional axisymmetric elements as shown in Figs. 1-3. [34-39].

The FE is applied to the 1-dimensional cylindrical coordinates heat transfer problem. The FE formulation is developed with the Galerkin Weighted-residual method. The appropriate working expressions of the conductance matrix, capacitance matrix and thermal load matrix are derived in details. The time dependent solution is obtained by applying the Backward Difference Schem.

#### Meshing the engineering problem of the domain

Let us consider a cylindrical molybdenum steel bar as shown in Fig. 1 which had been heated and then submerged in a cool quenching medium (sea water).



*Fig. 1 The axisymmetric one dimensional line (radius) element from the domain, on the cylindrical molybdenum steel bar which had been heated and then quenched in sea water.*

1-dimensional axisymmetric line (radius) element has been selected on this mathematical model. A 1-dimensional element in this work is defined by 5 nodes. Therefore, we can represent the variation of the dependent variable, such as the temperature history consequently the hardness at any point (node) even inside the heat treated quenched molybdenum steel bar.

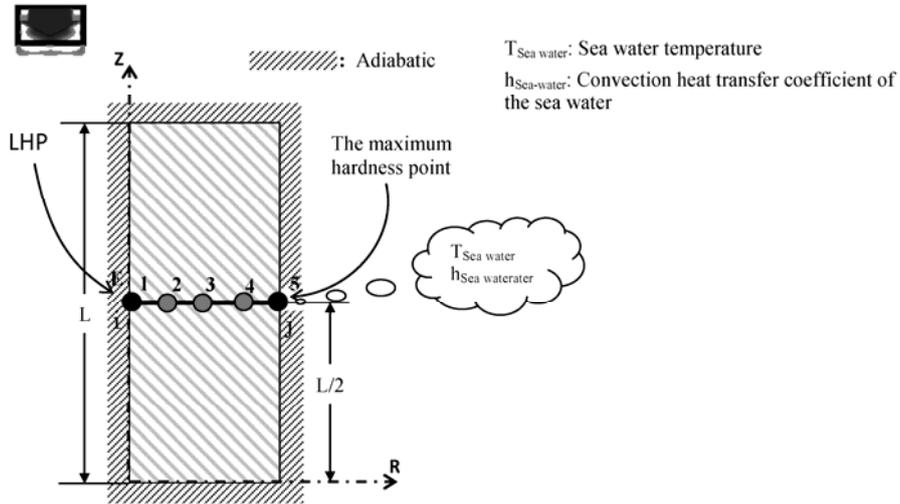


Fig. 2 The axisymmetric one dimensional line (radius) element from the domain on the axisymmetric rectangular cross-section.

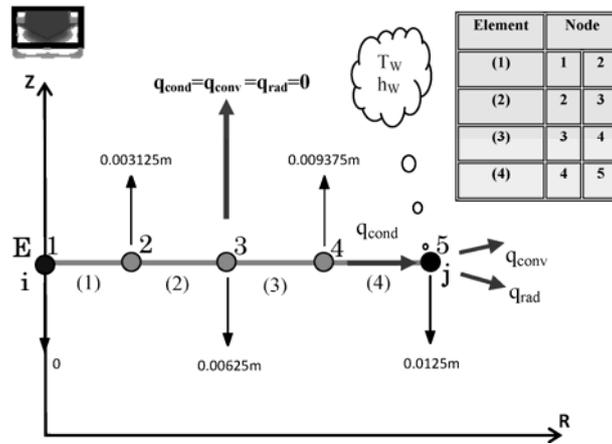


Fig. 3 The axisymmetric 1-dimensional line (radius) element from the domain shows the selected 4 elements with 5 nodes and the boundary at node j [5] for an element 4.

The linear temperature distribution for an element (radius) line, T is given by:

$$T^{(R)} = a_1 + a_2 R \tag{1}$$

where:  $T^{(R)}$  = nodal temperature as the function of R;  $a_1$  and  $a_2$  are constants, R is any point on the (radius) line element

Shape function of 1- dimensional axisymmetric element

The shape functions were to represent the variation of the field variable over the element. The shape function of axisymmetric 1-dimensional line (radius) element expressed in terms of the r coordinate and its coordinate are shown in Fig. 4; [34-39].

Global coordinate system being used to get the constants  $a_1$  and  $a_2$  thereby we can obtain the required shape function according to the below equations, which will be required for the Galerkin finite element solution.

$$S_i = \left( \frac{R_j - R}{R_j - R_i} \right) = \left( \frac{R_j - R}{L} \right) \quad (2a)$$

$$S_j = \left( \frac{R - R_i}{R_j - R_i} \right) = \left( \frac{R - R_i}{L} \right) \quad (2b)$$

Thus the temperature distribution of 1-dimensional radius for an element in terms of the shape function can be written as:

$$T^{(R)} = S_i T_i + S_j T_j = S^{(r)} \{T\} \quad (3)$$

Where  $[S^{(r)}] = [S_i \quad S_j]$  is a row vector matrix and

$\{T\} = \begin{Bmatrix} T_i \\ T_j \end{Bmatrix}$  is a column vector of nodal temperature of the element.

Eq. (3) can also be expressed in matrix form as:

$$T_{(R)} = [S_i \quad S_j] \begin{Bmatrix} T_i \\ T_j \end{Bmatrix} \quad (4)$$

Thus for 1-dimensional element we can write in general:

$$\Psi^{(e)} = [S_i \quad S_j] \begin{Bmatrix} \Psi_i \\ \Psi_j \end{Bmatrix} \quad (5)$$

Where  $\Psi_i$  and  $\Psi_j$  represent the nodal values of the unknown variable which in our case is temperature. The unknown parameters can be deflection, velocity etc.

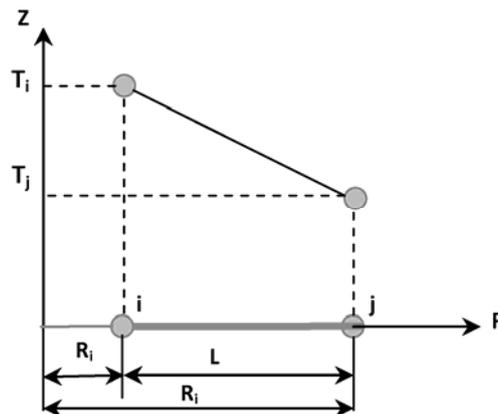


Fig. 4. 1-dimensional linear temperature distributions for an element (radius) line in Global Coordinate system.

### Natural area coordinate

Using the natural length coordinates and their relationship with the shape function by simplification of the integral of Galerkin solution:

The two length natural coordinates  $L_1$  and  $L_2$  at any point  $p$  inside the element are shown in Fig. 5 from which we can write:

$$L_1 = \frac{R_j - R}{R_j - R_i} = \frac{l_1}{L} \quad (6a)$$

$$L_2 = \frac{R - R_i}{R_j - R_i} = \frac{l_2}{L} \quad (6b)$$

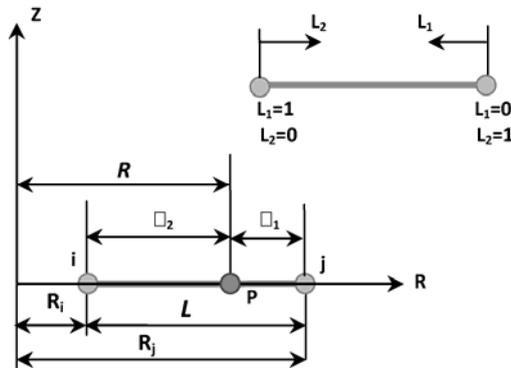


Fig. 5. Two-node line element showing interior point  $p$  and the two natural coordinates  $L_1$  and  $L_2$ .

Since it is a 1-dimensional element, there should be only one independent coordinate to define any point  $P$ . This is true even with natural coordinates as the two natural coordinates  $L_1$  and  $L_2$  are not independent, but are related as:

$$L_1 + L_2 = 1 \quad \text{or} \quad L_1 + L_2 = \frac{l_1}{L} + \frac{l_2}{L} = 1 \quad (7)$$

The natural coordinates  $L_1$  and  $L_2$  are also the shape functions for the line element, thus:

$$S_i = \left( \frac{R_j - R}{R_j - R_i} \right) = \left( \frac{R_j - R}{L} \right) = L_1$$

$$S_j = \left( \frac{R - R_i}{R_j - R_i} \right) = \left( \frac{R - R_i}{L} \right) = L_2$$

$$S_i = L_1, S_j = L_2, \quad (8)$$

$$R = R_j L_2 + R_i L_1 = R_i S_i + R_j S_j \quad (9)$$

$$\frac{\partial [S]_i}{\partial r} = \frac{\partial L_1}{\partial r} = \frac{-1}{R_j - R_i} = -\frac{1}{L} \quad (10)$$

$$\frac{\partial [S]_j}{\partial r} = \frac{\partial L_2}{\partial r} = \frac{1}{R_j - R_i} = \frac{1}{L} \quad (11)$$

Develop Equation for all Elements of the Domain

Derivation of equation of heat transfer in axisymmetric 1-dimensional line (radius) elements by applying the conservation of energy to a differential volume cylindrical segment has been done as shown in Fig. 6;

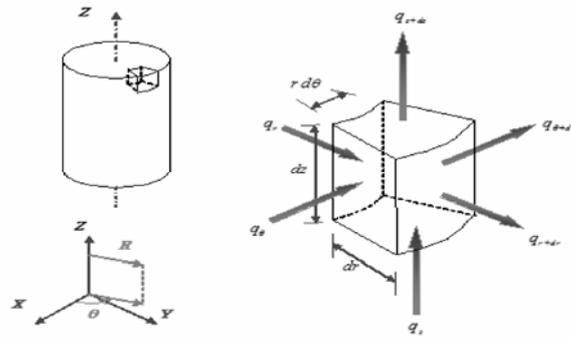


Fig. 6. Axisymmetric element from an axisymmetric body.

$$E_{in} - E_{out} + E_{generated} = E_{stored} \quad (12)$$

The transient heat transfer within the component during quenching can mathematically be described by simplifying the differential volume term [4, 11]; the heat conduction equation is derived and given by:

$$\frac{1}{r} \frac{d}{dr} \left( K_r r \frac{dT}{dr} \right) + \frac{1}{r^2} \frac{d}{d\theta} \left( K_\theta \frac{dT}{d\theta} \right) + \frac{d}{dz} \left( K_z \frac{dT}{dz} \right) + q = \rho c \frac{dT}{dt} \quad (13)$$

where:

$k_r$  = heat conductivity coefficient in  $r$ -direction,  $\text{W/m}\cdot^\circ\text{C}$ .

$k_\theta$  = heat conductivity coefficient in  $\theta$ -direction,  $\text{W/m}\cdot^\circ\text{C}$ .

$k_z$  = heat conductivity coefficient in  $z$ -direction,  $\text{W/m}\cdot^\circ\text{C}$ .

$T$  = temperature,  $^\circ\text{C}$ .

$q$  = heat generation,  $\text{W/m}^3$ .

$\rho$  = mass density,  $\text{kg/m}^3$ .

$c$  = specific heat of the medium,  $\text{J/kg}\cdot\text{K}$ .

$t$  = time, s.

#### The Necessary Assumptions Made to Solve the Problem

- i. For axisymmetric situations 1-dimensional line (radius) element, there is no variation of temperature in the  $Z$ -direction as shown in Figs. 1-3. This is because we have already assumed that in steel quenching and cooling process of the steel bar is insulated from convection at the cross section of the front and back.

It means that we have convection and radiation at one node only which is on the surface [node 5]. In our research we focus to calculate LHP which is at [node 1], i.e. the last point that will be cooled. This gives the maximum advantage to make our assumption safer, because it is the last point that will be affected by convection and radiation from the front and back cross section of the steel bar. Therefore we can write,

$$\left( \frac{\partial T}{\partial z} = 0 \right)$$

For axisymmetric situations, there is no variation of temperature in the  $\theta$ -direction, because it is clear from Figs. 1-3 that the temperature distribution along the radius will be the same if the radius move with angle  $\theta$ ,  $360^\circ$ . Therefore;

$$\left( \frac{\partial T}{\partial \theta} = 0 \right)$$

- ii. The thermal energy generation rate  $\dot{q}$  represents the rate of the conversion of energy from electrical, chemical, nuclear, or electromagnetic forms to thermal energy within the volume of the system. The conversion of the electric field will be studied in details in the second part of our research. Since in this manuscript no heat generation has been taken into account, therefore:

$$\dot{q} = 0$$

After simplifying, the Eq. (13) becomes:-

$$\frac{k}{r} \frac{\partial}{\partial r} \left( r \frac{\partial T}{\partial r} \right) = \rho c \frac{\partial T}{\partial t} \quad (14)$$

And also known as residual or partial differential equation:

$$\{R\} = \frac{k}{r} \frac{\partial}{\partial r} \left( r \frac{\partial T}{\partial r} \right) - \rho c \frac{\partial T}{\partial t} = 0 \quad (15)$$

#### Galerkin Weighted Residual Method Formulation

To suit the formulation into finite element analysis, Galerkin's weighted residual method is applied. The Galerkin residual in our case for 1-dimensional line (radius) element of unsteady state heat transfer can be obtained by integration the transpose of the shape functions times the residual which minimize the residual to zero becomes;

$$\int_V [S]^T \{R\}^{(e)} dv = 0 \quad (16)$$

Where,  $[S]^T$  = the transpose of the shape function matrix

$\{R\}^{(e)}$  = the residual contributed by element (e) to the final system of equations.

$$\frac{k}{r} \int [S]^T \frac{\partial}{\partial r} \left( r \frac{\partial T}{\partial r} \right) dv - \int [S]^T \rho c \frac{\partial T}{\partial t} dv = 0 \quad (17)$$

#### Chain rule

The term 1 and 2 of Eq. (17) can be re-arranged using the chain rule which states that:

$$(fg)' = fg' + gf'$$

Therefore,  $fg' = (fg)' - f'g$  then  $\frac{\partial}{\partial r} \left( [S]^T r \frac{\partial T}{\partial r} \right) = [S]^T \left\{ \frac{\partial}{\partial r} \left( r \frac{\partial T}{\partial r} \right) \right\} + r \frac{\partial T}{\partial r} \frac{\partial [S]^T}{\partial r}$

Term 1 of Eq. 17 is rearranged, thus:

$$[S]^T \left\{ \frac{\partial}{\partial r} \left( r \frac{\partial T}{\partial r} \right) \right\} = \frac{\partial}{\partial r} \left( [S]^T r \frac{\partial T}{\partial r} \right) - r \frac{\partial T}{\partial r} \frac{\partial [S]^T}{\partial r} \quad (18)$$

By substitute Eq. (18) into Eq. (17):

$$= \overbrace{\frac{k}{r} \int \left\{ \frac{\partial}{\partial r} \left( [S]^T r \frac{\partial T}{\partial r} \right) \right\} dv}^A - \overbrace{\frac{k}{r} \int \left\{ \frac{\partial [S]^T}{\partial r} r \frac{\partial T}{\partial r} \right\} dv}^B - \underbrace{\int [S]^T \left\{ \rho c \frac{\partial T}{\partial t} \right\} dv}_C \quad (19)$$

**Term A** is the heat convection term and contributes to the conductance and thermal load matrix. **Term B** is the heat conduction term and contributes to the conductance matrix. **Term C** is the transient equation and contributes to the capacitance matrix.

$$\text{Where: } \overbrace{\frac{k}{r} \int \frac{\partial}{\partial r} \left( [S]^T r \frac{\partial T}{\partial r} \right) dV}^A = \underbrace{-2\pi hrz [S]^T [S] [K] T^e}_{A_1} + \underbrace{2\pi hrz [S]^T T_f}_{A_2} - \underbrace{2\pi rz \varepsilon_s \sigma [S]^T [S] [K] T^e ([S] [K] T^e)^3}_{A_3} + \underbrace{2\pi rz \varepsilon_s \sigma [S]^T T_f^4}_{A_4}$$

Note that terms  $A_1$  and term  $A_3$  contributed to the conductance matrix since they contains the unknown temperature  $\{T\}$ . Terms  $A_2$  and  $A_4$  contributed to the thermal load matrix as  $T_f$  is the known fluid temperature. Terms  $A_3$  and term  $A_4$  refer to heat radiation and are very important if our heat treatment is Annealing [cooling in the furnace] or Normalizing [cooling in air or jet air], but they can be ignored if cooling is quenching in sea water as in our work.

From earlier explanations derivation and after simplification we can formulate the conductance matrix in the r-direction for term B. Finally we get:

**Term B** (the conduction term) contributes to the Conductance Matrix

$$\underbrace{\frac{k}{L} (R_j - R_i) \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} \begin{Bmatrix} T_i \\ T_j \end{Bmatrix}}_{K_c} \quad (20)$$

Similarly, term C, the unsteady state (transient) which contributes to the Capacitance Matrix, becomes:

**Term C** (heat stored) contributes to the Capacitance Matrix

$$\underbrace{\frac{L\rho c}{6} \begin{bmatrix} (3R_i + R_j) & (R_i + R_j) \\ (R_i + R_j) & (R_i + 3R_j) \end{bmatrix} \begin{Bmatrix} T_i \\ T_j \end{Bmatrix}}_C \quad (21)$$

### Term A (Heat Convection)

→ Term A<sub>1</sub> -Contributes to Conductance Matrix

Term A<sub>1</sub> (the convection term) contributes to the conductance matrix

$$\underbrace{-2hR_j \begin{bmatrix} 0 & 0 \\ 0 & 1 \end{bmatrix}}_{K_h} \begin{Bmatrix} T_i \\ T_j \end{Bmatrix} \quad (22)$$

→ Term A<sub>2</sub> -Contributes to Thermal Load Matrix

Term A<sub>2</sub> (the convection term) contributes to thermal load matrix

$$\underbrace{2hR_j T_w \begin{bmatrix} 0 \\ 1 \end{bmatrix}}_{f_h} \quad (23)$$

### Construct the element Matrices to the Global Matrix

The global, conductance, capacitance and thermal load matrices and the global of the unknown temperature matrix for all the elements in the domain are assembled i.e. the element's conductance; capacitance and thermal load matrices have been derived. Assembling these elements is necessary in all FE analysis.

Constructing these elements will result into the following FE equation:

$$[K]^{(G)} \{T\}^{(G)} + [C]^{(G)} \{\dot{T}\}^{(G)} = \{F\}^{(G)} \quad (24)$$

where:

$[K]=[Kc]+[Kh]$ : is conductance matrix due to Conduction (Elements 1 to 4) and heat loss through convection at the element's boundary (element 4 node 5) as shown in Figs. 1-3.

$\{T\}$ : is temperature value at each node, °C.

$[C]$ : is capacitance matrix, due to transient equation (heat stored)

$\{\dot{T}\}$ : is temperature rate for each node, °C/s.

$\{F\} = \{F_h\} + \{F_q\}$ : is heat load due to heat loss through convection at the element's boundary (element 4 node 5) and internal heat generation (element 4 node 5).

### Euler's method

Two point recurrence formulas will allow us to compute the nodal

temperatures as a function of time. In this paper, Euler's method which is known as the backward difference scheme (FDS) will be used to determine the rate of change in temperature, the temperature history at any point (node) of the steel bar [4, 11-14].

If the derivative of T with respect to time t is written in the backward direction and if the time step is not equal to zero ( $\Delta t \neq 0$ ), then we have:

$$\left\{ \dot{T} \right\} \approx \left\{ \frac{T(t) - T(t - \Delta t)}{\Delta t} \right\} \quad (25)$$

With;

$\{ \dot{T} \}$  = temperature rate ( $^{\circ}\text{C/s}$ ); T (t) = temperature at (t) s ( $^{\circ}\text{C}$ ); T (t -  $\Delta t$ ) = temperature at (t -  $\Delta t$ ) s, ( $^{\circ}\text{C}$ )

$\Delta t$  = selected time step (s) and t = time (s) (at starting time, t = 0)

By substituting the value of  $\{ \dot{T} \}$  into the finite element global equation, we obtain:

$$[K]^{(G)} \{T(t)\}^{(G)} + [C]^{(G)} \left\{ \frac{T(t) - T(t - \Delta t)}{\Delta t} \right\}^{(G)} = \{F(t)\}^{(G)} \quad (26)$$

Finally, the matrices become;

$$[[K]^{(G)} \Delta t + [C]^{(G)}] [T]_{j+1}^{(G)} = [C]^{(G)} \{T\}_i^{(G)} + \{F\}_{j+1}^{(G)} \Delta t \quad (27)$$

From Eq. (27) all the right hand side is completely known at time t, including t = 0 for which the initial condition apply.

Therefore, the nodal temperature can be obtained for a subsequent time given the temperature for the preceding time.

Once the temperature history is known the important mechanical properties such as hardness and strength of the molybdenum steel bar can be obtained.

## Application

### Calculation the temperature history

The present developed mathematical model is programmed using MATLAB to simulate the results of the temperature distribution with respect to time in transient state heat transfer of the industrial quenched molybdenum steel bar.

The cylindrical molybdenum steel bar has been heated to  $850^{\circ}\text{C}$ . Then being quenched in sea water with  $T_{\text{sea-water}} = 32^{\circ}\text{C}$  and convection heat transfer coefficient,  $h_{\text{sea-water}} = 1250 \text{ W/m}^2 \cdot ^{\circ}\text{C}$  [15]. The temperature history for the selected nodes of the cylindrical molybdenum steel bar after quenching is being identified in Figs. 7 and 8. The cylindrical bar was made from molybdenum steel, with properties as mentioned below.

Thermal capacity,  $\rho c$  (J/m<sup>3</sup>·°C):

$$0 \leq T \leq 650 \text{ } ^\circ\text{C}, \rho c = (0.004 T + 3.3) \times 10^6, 650 < T \leq 725 \text{ } ^\circ\text{C},$$

$$\rho c = (0.068 T - 38.3) \times 10^6$$

$$725 < T \leq 800 \text{ } ^\circ\text{C}, \rho c = (-0.086 T + 73.55) \times 10^6, T > 800 \text{ } ^\circ\text{C},$$

$$\rho c = 9.55 \times 10^6$$

Thermal conductivity,  $k$  (W/m·°C):

$$0 \leq T \leq 900 \text{ } ^\circ\text{C}, k = -0.022 T + 48, T > 900 \text{ } ^\circ\text{C}, k = 28.4$$

In our case Eq. (27) becomes:

$$[K]^{(G)} \{T\}^{(G)} + [C]^{(G)} \{\dot{T}\}^{(G)} = \{F\}^{(G)}$$

And their respective equations:

$$[K]^{(G)} = [K_c]^{(1)} + [K_c]^{(2)} + [K_c]^{(3)} + [K_c]^{(4)} + [K_h]^{(4)} \tag{28}$$

$$[C]^{(G)} = [C]^{(1)} + [C]^{(2)} + [C]^{(3)} + [C]^{(4)} \tag{29}$$

$$\{F\}^{(G)} = \{F_h\}^{(4)} \tag{30}$$

With the input data and boundary conditions provided, a sensitivity analysis is carried out with the developed program to obtain the temperature distribution at any point (node) of the quenched steel bar. As an example, is the transient state temperature distribution results of the selected five nodes from the center [W<sub>1</sub>] to the surface [W<sub>5</sub>] of the quenched steel bar which were computed as shown in Figs. 7 and 8.

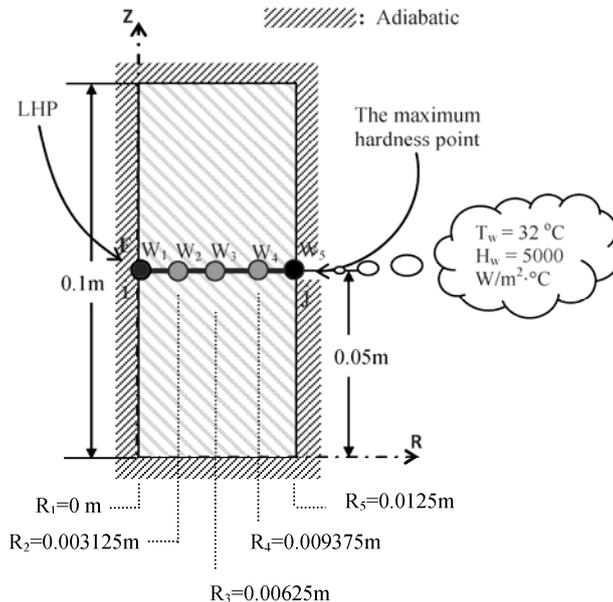


Fig. 7 The axisymmetric 1-dimensional line (radius) element from the domain when the radius equals 12.5mm, the selected 4 elements with 5 nodes and the boundary at node j [5] for an element 4.

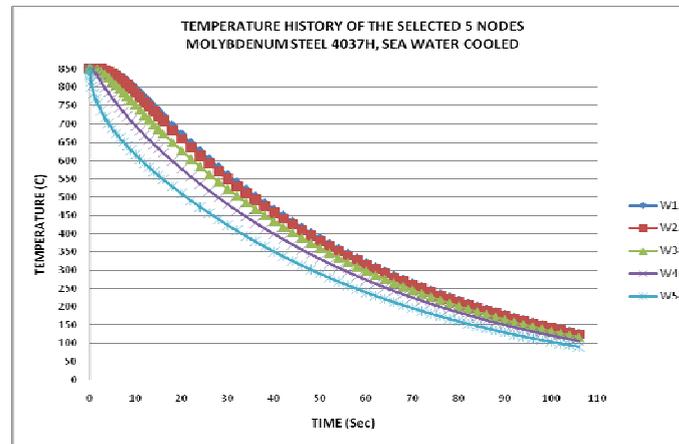


Fig. 8 Graph of temperature history along WW cross-section from MATLAB program.

#### LHP Calculation

##### Calculating the cooling time required

In this study, we choose to calculate the cooling time between 800°C and 500°C [11, 20-26]. Where, the characteristic cooling time, relevant for phase transformation in most structural steels is the time of cooling from 800 to 500°C (time  $t_{8/5}$ ) [7-10, 27-34].

$$t_c = t_{800} - t_{500}$$

From Fig. 8 we can determine the time taken for node  $W_1$  to reach 800°C,

$$t_{800} = 9.7 \text{ s.}$$

By the same way the time taken for node  $W_1$  to reach 500°C is

$$t_{500} = 36.3 \text{ s.}$$

So the Cooling time  $t_c$  for node  $W_1$ ;

$$t_c = t_{500} - t_{800} = 36.3 - 9.7 = 26.60 \text{ s.}$$

For nodes  $W_2$  to  $W_5$ , the cooling time  $t_c$  was calculated by the same way, the final results are shown in Table 1.

##### Calculating the Jominy distance from Standard Jominy distance versus cooling time curve

Obtained cooling time,  $t_c$ , will now be substituted into the Jominy distance versus cooling time curve in order to obtain the correspondent Jominy distance. Jominy distance can also be calculated by using polynomial expressions via polynomial regression via Microsoft Excel.

In this paper the standard Table [Cooling rate at each Jominy distance (Chandler, H., 1998)] will be used [33]. Then Jominy distance of nodes  $W_1$  to  $W_5$  will be calculated by using the data from [Cooling rate at each Jominy distance (Chandler, H., 1998)], the final results are shown in Table 1, where the rate of cooling (ROC) was defined by:

$$\text{ROC} = \frac{800^{\circ}\text{C} - 500^{\circ}\text{C}}{t_c} = \frac{800^{\circ}\text{C} - 500^{\circ}\text{C}}{t_{500^{\circ}\text{C}} - t_{800^{\circ}\text{C}}} \text{ (}^{\circ}\text{C/s)}$$

#### Predict the hardness of the quenched steel bar

The Rockwell hardness C (HRC) of molybdenum steel AISI-SAE 4037H can be calculated by using the relation between the Jominy distance and the HRC from the Practical date Handbook, the Timken Company 1835 Duebex Avenue SW Canton, Ohio 44706-2798 1-800-223. The final results are shown in Fig. 9 and Table 1.

Table 1 Cooling time, cooling rate, Jominy distance and HRC for the nodes  $W_1$  to  $W_5$ , sea water cooled.

Node	$t_c$ (s)	ROC ( $^{\circ}\text{C/s}$ )	Jominy-distance (mm)	Hardness (HRC)
$W_1$	26.61	11.273957	16.735	23.909
$W_2$	26.571	11.290504	16.717	23.919
$W_3$	26.183	11.457816	16.479	24.066
$W_4$	24.546	12.221950	15.639	24.652
$W_5$	20.466	14.658457	14.159	25.624

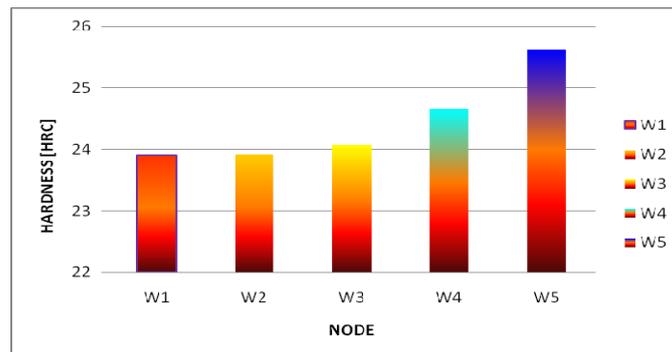


Fig. 9. Hardness distribution along WW cross-section for the nodes  $W_1$  to  $W_5$  from the centre to the surface, respectively, at half the length at the centre of the quenched steel bar.

#### Mathematical verification

The same data input for the steel properties and boundary condition used in the mathematical model was applied to the ANSYS software to verify the temperature simulation results. The temperature distribution from the ANSYS analysis is depicted figuratively as shown in Figs. 10.1a and 10.1b.

Fig. 10.1a shows the temperature distribution just before the steel bar becomes completely cooled whereas Fig. 10.1b shows the temperature distribution at the moment when the entire steel bar becomes completely cooled after 1498s.

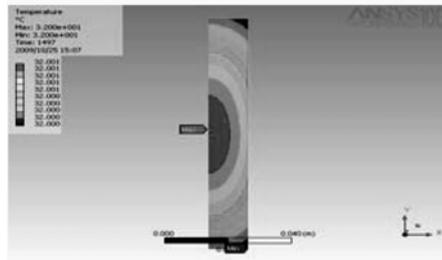


Fig. 10.1a

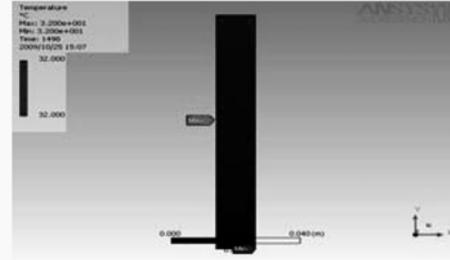


Fig.10.1b

The temperature time graph from the ANSYS analysis is depicted as shown in Fig. 11.

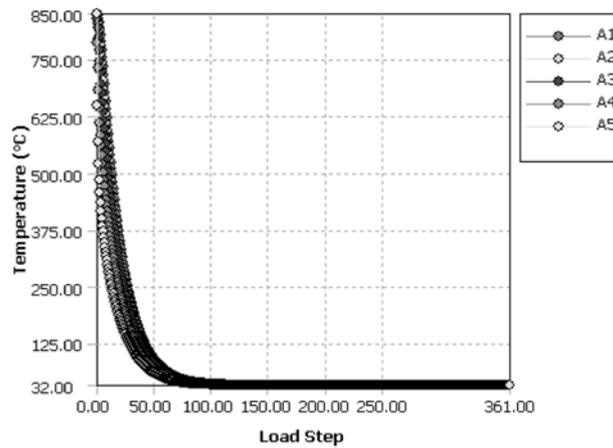


Fig. 11 Temperature-time graph from ANSYS

From the graphs shown in Fig. 8 (applied mathematical model) and Fig. 11 (applied ANSYS), it can be clearly seen that the temperature history of the quenched steel bar has the same pattern. The heat transfer across the steel bar is uniform. From Fig. 11 the cooling time, Jominy-distance and consequently the hardness of the quenched molybdenum steel 4037H at any point (node), even the (LHP) are also determined by ANSYS. The final results are shown in Table 2 and Fig. 12.

Table 2 Cooling time, cooling rate, Jominy distance and HRC for the nodes A<sub>1</sub> to A<sub>5</sub>, determined by ANSYS.

Node	Cooling time,	Cooling rate	J-distance (mm)	HRC
A <sub>1</sub>	32.112284	9.34222	18.788	22.668
A <sub>2</sub>	32.030048	9.36621	18.763	22.684
A <sub>3</sub>	31.466551	9.53393	18.590	22.795
A <sub>4</sub>	28.484752	10.53195	17.677	23.380
A <sub>5</sub>	24.074572	12.46128	15.468	24.759

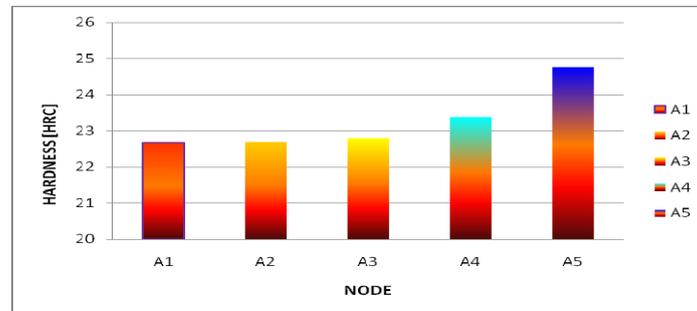


Fig. 12 Hardness distribution by ANSYS along AA cross section for the nodes  $A_1$  to  $A_5$  from the centre to the surface respectively at half the length at the centre of the quenched molybdenum steel bar.

From our results we found that in the mathematical model for the first node with  $W_1$  in the center, we found that HRC = 23.909. While in ANSYS for the same node  $A_1$ , we found that HRC = 22.668.

And for the nodes on the surfaces  $W_5$  and  $A_5$ , it was found that HRC = 25.624 and 24.759 for the mathematical model and ANSYS respectively. From the above, it can be seen that there is a strong agreement between both results. The difference between all the results of the mathematical model and the ANSYS simulations can be accounted due to the fact that the ANSYS software is for the commercial purpose, and thereby has some automated input data. But the developed mathematical model is precisely for a circular steel bar axisymmetric cross section. However, there is strong agreement between both results and thereby the result is validated where, the comparison indicated reliability of the proposed model. Also the results showed that the node on the surface will be the first which was completely cooled after quenching because it was in the contact with the cooling medium. The other nodes on the radial axis to the centre were successively cooled. The last point that would be completely cooled will be located at half the length of the centre. Hence, LHP will be at half the length of the centre of the quenched industrial molybdenum steel bar. It will be more important to know LHP once the radius of the quenched steel bar is large because LHP will be low. In other words, it will be lower than the hardness at the surface. This means that increasing the radius of the bar is inversely proportional with LHP. LHP calculation experimentally is an almost impossible task using manual calculation techniques also the earlier methods only used hardness calculated at the surface, which is higher than LHP, but this might have negative consequences resulting to deformation and failure of the component.

### Conclusion

A mathematical model of steel quenching has been developed to compute LHP of the quenched molybdenum steel 4037H at any point (node) in a specimen with cylindrical geometry. The model is based on the finite element Galerkin residual method. The numerical simulation of quenching consisted of numerical simulation of temperature transient field of cooling process. This mathematical model was verified

and validated by comparing the hardness results with ANSYS software simulations. From the mathematical model and ANSYS results, it is clear that the nodes at the surface [ $W_5$  and  $A_5$ ], respectively, cool faster than the nodes at the center [ $W_1$  and  $A_1$ ], respectively, because  $t_{CW_5}$  is less than  $t_{CW_1}$  and  $t_{CA_5}$  is less than  $t_{CA_1}$ . This means that the mechanical properties such as hardness will be different, i.e. the hardness at the surface nodes [ $W_5$  and  $A_5$ ] will be higher than the hardness at the center nodes [ $W_1$  and  $A_1$ ].

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